SHEAR WAVES IN AN ELASTIC WEDGE†

J. D. ACHENBACH

Department of Civil Engineering, Northwestern University, Evanston, Illinois

Abstract—An elastic wedge of interior angle $\varkappa \pi$ is subjected to spatially uniform but time-dependent shear tractions, which are applied to one or both faces of the wedge, parallel to the line of intersection of the faces. The transient wave propagation problem is solved by taking advantage of the dynamic similarity which characterizes problems without a fundamental length in the geometry. The shear stress $\tau_{\theta z}$ is evaluated, and it is found that the singularity near the vertex of the wedge is of the form $r^{(1/\kappa)-1}/(1 - \varkappa)$. The results show that the stress is not singular for interior angles less than π . As a special case we obtain the dynamic shear stress generated by the sudden opening of a semi-infinite crack in a homogeneously sheared unbounded medium.

INTRODUCTION

PROBLEMS that are concerned with the propagation of small deformations in linearly elastic solids are generally solved by means of Fourier transform techniques. If the region in space is unbounded, and if no characteristic length of the geometry enters the formulation, it may reasonably well be expected that a closed-form solution can be worked out. In obtaining this solution the use of Fourier transforms becomes, however, less attractive for more complicated regions in space, such as wedges. In this paper we consider, therefore, an alternative method of solution which is based on the dynamic similarity that characterizes problems without a fundamental length. This method, which has been used extensively in supersonic aerodynamics [1, 2], was applied by Miles to wave propagation problems in homogeneous elastic solids [3].

We consider the transient waves generated by spatially uniform but time dependent shear tractions which are applied to one face of a wedge in a direction parallel to the line of intersection of the faces. The other face is assumed free of surface tractions. Once the solution to this problem has been obtained we can, for arbitrary vertex angles, easily construct the solution for the cases where both faces are subjected to shear tractions, or where one face is clamped. As an interesting special solution we obtain the dynamic shear stress generated by the sudden opening of a semi-infinite shear crack in an unbounded medium. For the general case special attention is devoted to the singularity of the shear stress near the vertex of the wedge.

The transient diffraction of plane waves by a wedge in an acoustic medium was treated earlier by essentially the same method by Miles [4] and Keller and Blank [5].

⁺ The results of this paper were obtained in the course of research sponsored under Contract No. N00014-67-A-0109-0003, Task NR 064-496 by the Office of Naval Research, Washington, D.C., while the author was Visiting Associate Professor in the Department of the Aerospace and Mechanical Engineering Sciences, University of California at San Diego, La Jolla, California.

J. D. ACHENBACH

GOVERNING EQUATIONS

A homogeneous, isotropic, elastic wedge of interior angle $\varkappa \pi$, see Fig. 1, whose faces are defined by $\theta = 0$ and $\theta = \varkappa \pi$, respectively, is subjected on the face $\theta = 0$ to a uniform but time dependent shear traction $\tau_{\theta z}$. For the time being we assume $\varkappa \ge \frac{1}{2}$; the case $\varkappa < \frac{1}{2}$ will be considered later. The shear traction generates horizontally polarized shear motion in the z-direction only, which is governed by the equation

$$\frac{1}{r}\frac{\partial}{\partial r}\left(r\frac{\partial w}{\partial r}\right) + \frac{1}{r^2}\frac{\partial^2 w}{\partial \theta^2} = \frac{1}{c^2}\frac{\partial^2 w}{\partial t^2},\tag{1}$$

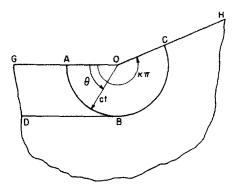


FIG. 1. Wavefronts at time t.

where w is the displacement in the z-direction, $c = (\mu/\rho)^{\frac{1}{2}}$ is the velocity of shear waves, and r, θ , z are cylindrical coordinates. It is assumed that the wedge is at rest prior to t = 0

$$t < 0, \qquad w(r, \theta, t) = \dot{w}(r, \theta, t) \equiv 0.$$
⁽²⁾

The displacement field generated by a uniform surface traction of arbitrary time dependence can be obtained by Duhamel superposition, once the displacements for a surface traction varying with time as the Dirac delta function have been found. We thus first consider the following boundary conditions

$$\theta = 0, \quad r \ge 0: \qquad \tau_{\theta z} = \frac{\mu}{r} \frac{\partial w}{\partial \theta} = \tau_1 \delta(t)$$
 (3)

$$\theta = \varkappa \pi, \quad r \ge 0: \qquad \tau_{\theta z} = \frac{\mu}{r} \frac{\partial w}{\partial \theta} \equiv 0.$$
 (4)

The problem at hand consists of finding a solution of equation (1) satisfying the initial conditions (2) and the boundary conditions (3) and (4).

Some observations on the pattern of waves propagating in the wedge can be deduced from elementary principles of wave propagation. The surface traction (3) generates a plane wave with constant displacement

$$w_1 = -\frac{c\tau_1}{\mu}.$$
 (5)

This wave is called the primary wave, and in Fig. 1 its wavefront at an arbitrary time t is indicated by BD. Since the wedge is at rest prior to time t = 0, the medium is undisturbed ahead of the wavefront BD, and as discussed above the displacement is constant behind it. In addition to the primary wave, the vertex of the wedge, as well as the non-uniformity of the surface traction across the vertex, generates a cylindrical wave with center at O. Since the displacement is continuous across the cylindrical wavefronts we have for $\varkappa \geq \frac{1}{2}$: $\omega \equiv 0$ along BC.

There is no characteristic length in the problem, and it is thus to be expected that the solution shows dynamic similarity, i.e. the displacement is a function of r/t and θ . It is then expedient to introduce as a new independent variable

$$s = r/t. (6)$$

As a function of s and θ the displacement w(s, θ) satisfies

$$s^{2}\left(1-\frac{s^{2}}{c^{2}}\right)\frac{\partial^{2}w}{\partial s^{2}}+s\left(1-\frac{2s^{2}}{c^{2}}\right)\frac{\partial w}{\partial s}+\frac{\partial^{2}w}{\partial \theta^{2}}=0.$$
(7)

For s < c, equation (7) is elliptic. Upon introducing Chaplygin's transformation

$$\beta = -\cosh^{-1}\left(\frac{c}{s}\right),\tag{8}$$

equation (7) reduces to

$$\frac{\partial^2 w}{\partial \beta^2} + \frac{\partial^2 w}{\partial \theta^2} = 0.$$
(9)

The solution of Laplace's equation may be written as the real part of an analytic function $\chi(\beta, \theta)$,

$$w(\beta, \theta) = \operatorname{Re} \chi(\beta, \theta).$$
 (10)

For s > c, equation (7) is hyperbolic and may be reduced to the canonical form by the transformation $s = c \sec \alpha$.

From Fig. 1 the region in which equation (7) is elliptic is now identified as the cylindrical region ABC. All that now remains to be done is to find a harmonic function $w(\beta, \theta)$ in the segment $0 \le \theta \le \varkappa \pi$, $s \le c$ satisfying boundary conditions which for $\varkappa \ge \frac{1}{2}$ take the form

$$\theta = 0, \quad s \le c: \quad \frac{\partial w}{\partial \theta} = 0$$
 (11)

$$s = c, \quad 0 \le \theta \le \frac{\pi}{2}; \quad w = w_1 = -\frac{c\tau_1}{\mu}$$
 (12)

$$s = c, \qquad \frac{\pi}{2} \le \theta \le \varkappa \pi: \qquad w = 0$$
 (13)

$$\theta = \varkappa \pi, \quad s \le c: \quad \frac{\partial w}{\partial \theta} = 0.$$
(14)

In equation (11) we have used that for t > 0 the surface traction has returned to zero, see equation (3). The function $w(\beta, \theta)$ can be obtained in several ways; here we elect to map the segment $s \le c$, $0 \le \theta \le \kappa \pi$ on the half-plane $\eta \ge 0$ by a conformal mapping which was introduced by Craggs [2]

$$\zeta = \xi + i\eta = \operatorname{sech}[(\beta + i\theta)/\varkappa].$$
(15)

Equation (15) can also be written as

$$\zeta = \left[\cosh \frac{\beta}{\varkappa} \cos \frac{\theta}{\varkappa} + i \sinh \frac{\beta}{\varkappa} \sin \frac{\theta}{\varkappa} \right]^{-1}.$$
 (16)

The mapping of the segment $s \le c$, $0 \le \theta \le \varkappa \pi$ on the ζ -plane is shown in Fig. 2, where the positions of the various points are indicated. The boundary conditions (11)–(14) are converted into conditions on the real axis, and we find for $\varkappa \ge 1$

$$|\xi| \le 1: \qquad \frac{\partial w}{\partial \eta} = 0 \tag{17}$$

$$1 \le \xi \le 1/\cos\frac{\pi}{2\varkappa}; \qquad w = w_1 \tag{18}$$

$$\xi \ge 1/\cos\frac{\pi}{2\varkappa}; \qquad w = 0 \tag{19}$$

$$\xi \le -1: \qquad w = 0. \tag{20}$$

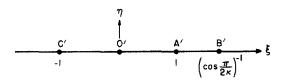


FIG. 2. The ζ -plane for $\varkappa \geq 1$.

For $1 \ge \varkappa \ge \frac{1}{2}$ the point B' is located on the negative real axis and (17)-(20) must be modified accordingly.

In this paper we are interested in the stresses, in particular the shear stress $\tau_{\theta z}$, which may be written in the form

$$\tau_{\theta_2} = \frac{\mu}{r} \operatorname{Re} \left[\frac{\mathrm{d}\chi}{\mathrm{d}\zeta} \frac{\partial\zeta}{\partial\theta} \right].$$
(21)

According to (17), $d\chi/d\zeta$ is real for $|\xi| \le 1$, and from (18)-(20) we conclude that $d\chi/d\zeta$ is imaginary elsewhere along the real axis, except at the point

$$\xi = \xi_B = 1/\cos\frac{\pi}{2\varkappa}.$$
(22)

At this point Re χ is discontinuous, and Re $d\chi/d\zeta$ is a delta function, which implies that $d\chi/d\zeta$ has a simple pole at $\zeta = \zeta_B$. An expression satisfying the foregoing requirements is

$$\frac{\mathrm{d}\chi}{\mathrm{d}\zeta} = \frac{i}{(\zeta^2 - 1)^{\frac{1}{2}}} \frac{B}{\zeta - \zeta_B}.$$
(23)

The constant B is found by integrating counterclockwise along a small semi-circle around $\zeta = \zeta_B$ and equating the result to $-w_1$. We obtain

$$B = \frac{w_1}{\pi} (\xi_B^2 - 1)^{\frac{1}{2}} = -\frac{c\tau_1}{\pi\mu} \tan \frac{\pi}{2\varkappa}.$$
 (24)

From equation (15) we compute

$$\frac{\partial \zeta}{\partial \theta} = -\frac{i}{\varkappa} \frac{\tanh[\beta/\varkappa + i(\theta/\varkappa)]}{\cosh[\beta/\varkappa + i(\theta/\varkappa)]}.$$
(25)

Upon substitution of (16) and (24) into (23), and subsequent substitution of (23) and (25) into (21) we obtain

$$\tau_{\theta z} = \frac{c\tau_1}{r} \frac{\sin(\pi/2\varkappa)}{\pi\varkappa} \operatorname{Re}\left\{i \left| \left[\cos\frac{\pi}{2\varkappa} - \cosh\left(\frac{\beta}{\varkappa} + i\frac{\theta}{\varkappa}\right) \right] \right\}.$$
(26)

The real part can easily be evaluated, and the result is

$$\tau_{\theta z} = -\frac{c\tau_1}{r} \frac{\sin(\pi/2\varkappa)}{\pi\varkappa} F_1\left(\frac{r}{ct},\theta\right),\tag{27}$$

where

$$F_1\left(\frac{r}{ct},\theta\right) = \frac{\sinh\left(\beta/\varkappa\right)\sin\left(\theta/\varkappa\right)}{\left[\left(\cos\pi/2\varkappa\right) - \left(\cosh\beta/\varkappa\right)\left(\cos\theta/\varkappa\right)\right]^2 + \left[\left(\sinh\beta/\varkappa\right)\left(\sin\theta/\varkappa\right)\right]^2}.$$
 (28)

Equation (27) represents the stress within the cylindrical wavefront, $r \le ct$. For $r \ge ct$ and $0 \le \theta \le \pi/2$ we have

$$\tau_{r\theta} = \tau_1 \cos \theta \,\delta \,\left(t - \frac{r}{c} \sin \theta\right). \tag{29}$$

It is of particular interest to investigate for t > 0 the singularity of $\tau_{\theta z}$ as $r \to 0$. To this end we evaluate asymptotic expressions for $\sinh(\beta/\varkappa)$ and $\cosh(\beta/\varkappa)$ for small r/ct. Using a well-known representation for \cosh^{-1} , we find from (8) and (6):

$$\beta = -\ln\left\{\frac{ct}{r} + \left[\left(\frac{ct}{r}\right)^2 - 1\right]^{\frac{1}{2}}\right\}.$$
(30)

Equation (30) is subsequently used to write

$$\cosh \frac{\beta}{\varkappa} = \frac{1}{2} \left[\left(\frac{r}{ct} \right)^{1/\varkappa} \left\{ 1 + \left[1 - \left(\frac{r}{ct} \right)^2 \right]^{\frac{1}{2}} \right\}^{-1/\varkappa} + \left(\frac{ct}{r} \right)^{1/\varkappa} \left\{ 1 + \left[1 - \left(\frac{r}{ct} \right)^2 \right]^{\frac{1}{2}} \right\}^{1/\varkappa} \right]$$
(31)
$$\sinh \frac{\beta}{\varkappa} = \frac{1}{2} \left[\left(\frac{r}{ct} \right)^{1/\varkappa} \left\{ 1 + \left[1 - \left(\frac{r}{ct} \right)^2 \right]^{\frac{1}{2}} \right\}^{-1/\varkappa} - \left(\frac{ct}{r} \right)^{1/\varkappa} \left\{ 1 + \left[1 - \left(\frac{r}{ct} \right)^2 \right]^{\frac{1}{2}} \right\}^{1/\varkappa} \right].$$
(32)

Thus for $(r/ct) \ll 1$

$$\cosh\frac{\beta}{\varkappa} \sim \left(\frac{ct}{r}\right)^{1/\varkappa} 2^{1/\varkappa - 1} \tag{33}$$

$$\sinh\frac{\beta}{\varkappa} \sim -\left(\frac{ct}{r}\right)^{1/\varkappa} 2^{1/\varkappa - 1}.$$
(34)

For $\varkappa \geq \frac{1}{2}$ the singularity in the shear stress is thus obtained as

$$\tau_{\theta z} \sim \frac{2c\tau_1}{\pi \varkappa} (2ct)^{-1/\varkappa} \sin\left(\frac{\pi}{2\varkappa}\right) \sin\left(\frac{\theta}{\varkappa}\right) r^{1/\varkappa - 1}.$$
 (35)

We note that the singularity vanishes for $\theta = 0$ and $\theta = \varkappa \pi$, and reaches a maximum for $\theta = \varkappa \pi/2$. The singularity disappears altogether if $\varkappa \le 1$. The shear stress is thus singular only if the interior angle of the wedge exceeds π .

BOTH FACES SUBJECTED TO SHEAR TRACTIONS

If the faces defined by $\theta = 0$ and $\theta = \varkappa \pi$ are subjected to shear tractions $\tau_{\theta z} = \tau_1 \delta(t)$ and $\tau_{\theta z} = \tau_2 \delta(t)$, respectively, the shear stress $\tau_{\theta z}$ is obtained by simple superposition. We obtain for $r \le ct$

$$\tau_{\theta z} = -\frac{c}{r} \frac{\sin(\pi/2\varkappa)}{\pi\varkappa} \left\{ \tau_1 F_1\left(\frac{r}{ct}, \theta\right) + \tau_2 F_2\left(\frac{r}{ct}, \theta\right) \right\},\tag{36}$$

where $F_1(r/ct, \theta)$ is defined by (28), and

$$F_2\left(\frac{r}{ct},\theta\right) = \frac{\sinh(\beta/\varkappa)\sin(\theta/\varkappa)}{\left[(\cos \pi/2\varkappa) + (\cosh \beta/\varkappa)(\cos \theta/\varkappa)\right]^2 + \left[(\sinh \beta/\varkappa)(\sin \theta/\varkappa)\right]^2}.$$
(37)

In addition, we have for $r \ge ct$ and $0 \le \theta \le \pi/2$ the plane wave (29), and for $\varkappa \pi \ge \theta \ge \varkappa \pi - \pi/2$ we have

$$\tau_{\theta z} = \tau_2 \cos(\varkappa \pi - \theta) \quad \delta \left[t - \frac{r}{c} \sin(\varkappa \pi - \theta) \right]. \tag{38}$$

It is clear that for this case the singularity is of the form

$$\tau_{\theta z} \sim \frac{2c(\tau_1 + \tau_2)}{\pi \varkappa} (2ct)^{-1/\varkappa} \sin\left(\frac{\pi}{2\varkappa}\right) \sin\left(\frac{\theta}{\varkappa}\right) r^{1/\varkappa - 1}.$$
(39)

Thus, the singularity vanishes if the two shear tractions are of the same magnitude and sense, i.e. if $\tau_1 = -\tau_2$.

THE SHEAR STRESS FOR STEP SURFACE TRACTIONS

It is of considerable physical interest to compute the dynamic stresses for the case that the surface traction varies as a Heaviside step function,

$$\theta = 0, \qquad r \ge 0: \qquad \tau_{\theta z} = T_1 \mathbf{1}(t) \tag{40}$$

$$\theta = \varkappa \pi, \qquad r \ge 0: \qquad \tau_{\theta z} = 0.$$
 (41)

For $0 \le \theta \le \pi/2$ the shear stress $\tau_{\theta z}$ is now obtained as the sum of the integrals of (27) and (29)

$$\tau_{\theta z} = T_1 \cos \theta \, \left(t - \frac{r}{c} \sin \theta \right) - \frac{c T_1}{r} \frac{\sin(\pi/2\varkappa)}{\pi\varkappa} \left(t - \frac{r}{c} \right) \int_{r/c}^t F_1\left(\frac{r}{cv}, \theta\right) dv. \tag{42}$$

For $\theta \ge \pi/2$ we obtain

$$\tau_{\theta z} = -\frac{cT_1}{r} \frac{\sin(\pi/2\varkappa)}{\pi\varkappa} \mathbf{1} \left(t - \frac{r}{c} \right) \int_{r/c}^t F_1 \left(\frac{r}{cv}, \theta \right) dv.$$
(43)

The behaviour of $\tau_{\theta z}$ for small values of r is found by introducing the new variable

$$p = \frac{cv}{r}.$$
(44)

The integrals in (42) and (43) then become

$$\tau_{\theta z} = -T_1 \frac{\sin(\pi/2\varkappa)}{\pi\varkappa} \mathbf{1} \left(\frac{ct}{r} - 1 \right) \int_1^{ct/r} F_1(p,\theta) \,\mathrm{d}p. \tag{45}$$

We are interested in approximating (45) for large values of ct/r. For large p the $\cosh[\beta(p)/\varkappa]$ and $\sinh[\beta(p)/\varkappa]$ may be approximated by, see (31) and (32),

$$\cosh\frac{\beta}{\varkappa} \sim p^{1/\varkappa} 2^{1/\varkappa - 1} \tag{46}$$

$$\sinh\frac{\beta}{\varkappa} \sim -p^{1/\varkappa} 2^{1/\varkappa - 1}. \tag{47}$$

The integral can now be evaluated, and for $\varkappa \geq \frac{1}{2}$, $\varkappa \neq 1$, and $r/ct \ll 1$ we find

$$\tau_{\theta z} \sim \frac{T_1}{\pi(\varkappa - 1)} \left(\frac{r}{2ct}\right)^{1/\varkappa - 1} \sin\left(\frac{\pi}{2\varkappa}\right) \sin\left(\frac{\theta}{\varkappa}\right). \tag{48}$$

For $\varkappa = 1$ and $r/ct \ll 1$ we obtain

$$\tau_{\theta z} \sim \frac{1}{\pi} T_1 \sin \theta \ln \left(\frac{ct}{r}\right).$$
 (49)

We note that no singularity occurs if the interior angle of the wedge is less than π .

SPECIAL CASES

The integrals in (42) and (43) can be evaluated with particular ease for three special values of $\kappa : \kappa = 0.5$, $\kappa = 1$ and $\kappa = 2$.

For $\varkappa = 0.5$ we have a quarter-space subjected to uniform surface tractions. From equations (27) and (42) it is immediately noted that the cylindrical wave vanishes, as it should, and we thus are left with just the plane primary wave.

The case $\varkappa = 1$ is concerned with a half-space. We assume uniform surface tractions that are of different magnitudes for $\theta = 0$ and $\theta = \pi$, respectively. Equation (36) then

reduces to

$$\tau_{\theta z} = -\frac{c}{r} \frac{1}{\pi} \frac{(\tau_1 + \tau_2) \sinh \beta \sin \theta}{\cosh^2 \beta - \sin^2 \theta}.$$
 (50)

Equations (31) and (32) become

$$\cosh\beta = \frac{ct}{r} \tag{51}$$

$$\sinh \beta = -\left[\left(\frac{ct}{r}\right)^2 - 1\right]^{\frac{1}{2}}.$$
(52)

Thus for $r \leq ct$:

$$\tau_{\theta z} = \frac{c}{\pi} \frac{(\tau_1 + \tau_2)(\sin\theta)[(ct)^2 - r^2]^{\frac{1}{2}}}{[(ct)^2 - r^2\sin^2\theta]}.$$
(53)

The cylindrical wave vanishes, of course, altogether for $\tau_1 = -\tau_2$. For $r/ct \ll 1$ we recover the behaviour shown in equation (39) for $\varkappa = 1$.

For step surface tractions the integration of (36) yields

$$I_{\theta z} = \frac{1}{2} \frac{c}{\pi r} (T_1 + T_2) \int_{br}^{t} \left\{ \frac{(s^2 - b^2 r^2)^{\frac{1}{2}}}{s - br \sin \theta} - \frac{(s^2 - b^2 r^2)^{\frac{1}{2}}}{s + br \sin \theta} \right\} \mathrm{d}s, \tag{54}$$

where we have introduced the slowness b,

$$b = \frac{1}{c}.$$
 (55)

This integral can be evaluated to yield for $br \leq t$

$$I_{\theta z} = \frac{T_1 + T_2}{\pi} \left\{ \sin \theta \ln \left[\frac{(t^2 - b^2 r^2)^{\frac{1}{2}} + t}{br} \right] - \frac{1}{2} \cos \theta \sin^{-1} \left[\frac{t \sin \theta - br}{t - br \sin \theta} \right] - \frac{1}{2} \cos \theta \sin^{-1} \left[\frac{t \sin \theta + br}{t + br \sin \theta} \right] \right\}.$$
(56)

For $\theta \leq \pi/2$ the shear stress is then obtained as

$$\tau_{\theta z} = T_1 \cos \theta \ \mathbf{1}(t - br \sin \theta) + I_{\theta z} \mathbf{1}(t - br). \tag{57}$$

For $\theta \ge \pi/2$ we have

$$\tau_{\theta z} = -T_2 \cos \theta \ \mathbf{1}(t - br \sin \theta) + I_{\theta z} \mathbf{1}(t - br). \tag{58}$$

For $br/t \ll 1$ the behaviour is as indicated by (49).

The case $\varkappa = 2$ corresponds to a semi-infinite slit in an unbounded medium subjected to different shear tractions on the two faces. For

$$\tau_1 = \tau_2 = \tau, \tag{59}$$

386

equation (36) reduces to

$$\tau_{\theta z} = -\frac{c}{r} \frac{2^{\frac{1}{2}}}{\pi} \frac{\tau \sinh \frac{1}{2} \beta \sin \frac{1}{2} \theta \left(1 + \cosh \beta + \cos \theta\right)}{(\cosh \beta)^2 - (\sin \theta)^2}.$$
 (60)

For $\varkappa = 2$, equation (32) becomes

$$\sinh \frac{1}{2}\beta = -2^{-\frac{1}{2}} \left(\frac{ct}{r} - 1\right)^{\frac{1}{2}}.$$
 (61)

After some further manipulation we then obtain for r < ct

$$\tau_{\theta z} = \frac{\tau}{\pi} 2^{-\frac{1}{2}} \left(\frac{ct}{r} - 1 \right)^{\frac{1}{2}} \left\{ \frac{\cos \frac{1}{2} \left(\pi/2 - \theta \right)}{t - (r/c) \sin \theta} - \frac{\sin \frac{1}{2} \left(\pi/2 - \theta \right)}{t + (r/c) \sin \theta} \right\}.$$
 (62)

The behaviour for $r/ct \ll 1$ agrees with what is obtained from equation (39) for $\varkappa = 2$.

The integrals to determine the stress for step surface tractions can again be evaluated explicitly, and we obtain for $br \le t$

$$\left(\frac{2r}{c}\right)^{\frac{1}{2}} \frac{\pi J_{\theta z}}{2T} = \left\{ (t - br)^{\frac{1}{2}} + [br(1 - \sin\theta)]^{\frac{1}{2}} \tan^{-1} \left[\frac{t - br}{br(1 - \sin\theta)} \right]^{\frac{1}{2}} \right\} \cos \frac{1}{2} \left(\frac{\pi}{2} - \theta \right) - \left\{ (t - br)^{\frac{1}{2}} + [br(1 + \sin\theta)]^{\frac{1}{2}} \tan^{-1} \left[\frac{t - br}{br(1 + \sin\theta)} \right]^{\frac{1}{2}} \right\} \sin \frac{1}{2} \left(\frac{\pi}{2} - \theta \right),$$
(63)

where T is the applied surface traction. For $0 \le \theta \le \pi/2$ and $2\pi \ge \theta \ge 3\pi/2$ the shear stress $\tau_{\theta z}$ is obtained as

$$\tau_{\theta z} = T \cos \theta \ \mathbf{1}(t - br \sin \theta) + J_{\theta z} \mathbf{1}(t - br). \tag{64}$$

For $\pi/2 \le \theta \le 3\pi/2$ we find

$$\tau_{\theta z} = J_{\theta z} \mathbf{1} (t - br). \tag{65}$$

If $r/ct \ll 1$ the angular behaviour is of the form

$$\frac{\pi\tau_{\theta z}}{2T} \sim \left(\frac{ct}{2r}\right)^{\frac{1}{2}} \left\{\cos\frac{1}{2}\left(\frac{\pi}{2}-\theta\right) - \sin\frac{1}{2}\left(\frac{\pi}{2}-\theta\right)\right\},\tag{66}$$

which agrees with (48).

Equations (64) and (65) when superposed on a uniform shear stress $\tau_{yz} = -T$ yield the solution for the stress $\tau_{\theta z}$ due to the sudden opening of a semi-infinite crack in an unbounded medium which was in a state of uniform shear prior to fracture. As is common in crack problems we find that the stress is singular near the crack tip as $r^{-\frac{1}{2}}$.

To obtain an independent check of equation (62), the problem of a semi-infinite slit in an unbounded medium subjected to shear tractions on the faces ($\varkappa = 2$) was also solved by means of Fourier transform techniques. By using the one-sided Laplace transform over time, and the exponential Fourier transform over one of the spatial coordinates, in conjunction with an application of the Wiener-Hopf technique, the transformed stress was obtained. The inverse transform was subsequently evaluated by means of Cagniard's method. The solution obtained by this rather cumbersome procedure verified equation (62). It appears from equation (66) that for step surface tractions the shear stress increases beyond bounds as t increases. This indicates that a static solution does not exist. A similar result was found by Maue, equations (72) and (76) of Ref. [6], for the problem of the slit subjected to normal surface tractions.

CONCLUDING REMARKS

Although only the case $\varkappa \ge 0.5$ was treated here, it is clear that wave propagation in wedges with interior angles less than $\pi/2$ can be treated by using superposition in conjunction with symmetry or antisymmetry properties. Thus if the face $\theta = \varkappa^* \pi (0.25 \le \varkappa^* < 0.5)$ is free of traction, and $\theta = 0$ is subjected to $\tau_1 \delta(t)$, we can simply use (36) with $\varkappa = 2\varkappa^*$, and $\tau_2 = -\tau_1$ to obtain the solution. If the face $\theta = \varkappa^* \pi$ is clamped, i.e. $w \equiv 0$, we substitute $\varkappa = 2\varkappa^*$ and $\tau_2 = \tau_1$ in equation (36). For $\varkappa^{**} < 0.25$ the solution for $0.25 \le \varkappa^* < 0.5$ has to be used in the just described procedure.

It is finally concluded that the method of homogeneous functions is an efficient method to study the dynamic response of an elastic wedge to spatially uniform shear tractions. For surface tractions varying in time as Heaviside step functions it was found that for interior angles $\varkappa \pi$ the singularity of the shear stress $\tau_{\theta z}$ is of the form $[r^{(1/\varkappa)-1}/(1-\varkappa)]$, which shows that the stress is singular only for $\varkappa \ge 1$.

REFERENCES

- [1] G. N. WARD, Linearized Theory of Steady High-Speed Flow, Chapter 7. Cambridge University Press (1955).
- [2] J. W. CRAGGS, The supersonic motion of an airfoil through a temperature front. J. fluid Mech. 3, 176 (1957).
- [3] J. W. MILES, Homogeneous solutions in elastic wave propagation. Q. appl. Math. 18, 37 (1959).
- [4] J. W. MILES, On the diffraction of an acoustic pulse by a wedge. Proc. R. Soc. A212, 543 (1952).
- [5] J. B. KELLER and A. BLANK, Diffraction and reflection of pulses by wedges and corners. The Theory of Electromagnetic Waves, edited by M. KLINE, p. 75. Dover (1965).
- [6] A. W. MAUE, Die Entspannungswelle bei plötzlichem Einschnitt eines gespannten elastischen Körpers. Z. angew. Math. Mech. 34, 1 (1954).

(Received 27 June 1969)

Абстракт—Упругий клин, внутренный угол которого составляет $x\pi$, находится под влиянием сил сдвига, пространственно однородных, но завцсящих от времени. Силы приложенные к одной или к двум сторонам клина и параллельны к клинам пересечения этих сторон. Решается задача по распространению поперечной волны, используя динамическое подобие, которое характеризует задачи без основной длины в геометрии. Определяется напряжение сдвига $\tau_{\theta z}$ и находится, что сингулярность близи вершины клина имеет форму $r^{(1/X)-1}/(1-x)$. Результаты показывают, что напряжения не сингулярны для внутренных углов менее π . В качестве специального случая, получаются динамические напряжения сдвига, вызванные внезапным откритием полу-бесконечной трещины в однородно подвергающейся сдвигам, безграничной среде.